

- 1 a) Climate sensitivity is a measure of how much global mean surface temperatures will change per unit change in radiative forcing. According to the IPCC, a doubling of CO_2 , which is associated with a change in radiative forcing of 4 W/m^2 , will produce a change in global mean surface temperatures of about 3°C . This is the most likely estimate of the change, but the likely range of the temperature change is between $2-4^\circ\text{C}$.
- b) Removing all O_3 above 30 km will increase the flux of UV radiation to the surface, which will result in a warming of the surface. The loss of UV absorption by stratospheric O_3 will result in a cooling of the stratosphere.
- c) Ozone absorbs IR at $9.6 \mu\text{m}$ and is an effective greenhouse gas in the upper troposphere. It is in the upper troposphere where the optical path becomes small enough that IR photons emitted by O_3 can make it out to space. Removing O_3 at these altitudes will allow radiation from lower (warmer) regions of the atmosphere to escape to space, cooling the surface. Removing O_3 in the lower troposphere has little impact on surface temperatures since O_3 there is absorbing and emitting radiation at nearly the same temperature as the surface.

2) Equilibrium will reflect a balance between

a) Solar radiation and emission from the satellite

$$\text{power absorbed} = L_0 \pi r^2$$

$$\text{power emitted} = 4\pi r^2 \sigma T^4$$

the albedo = 0 since it absorbs all radiation

$$L_0 \pi r^2 = 4\pi r^2 \sigma T^4$$

$$T = \left[\frac{L_0}{4\sigma} \right]^{1/4}$$

When the satellite goes into Earth's shadow, it will lose energy at the rate $4\pi r^2 \sigma T^4$

For mass $m = 10^3 \text{ kg}$ and specific heat c

$$\text{energy loss} = mc \frac{dT}{dt} = 4\pi r^2 \sigma T^4$$

$$\Rightarrow mc \frac{dT}{dt} = \pi r^2 L_0$$

$$\Rightarrow \frac{dT}{dt} = \frac{\pi r^2 L_0}{mc} = \frac{\pi (1\text{m})^2 (1370 \text{ W m}^{-2})}{10^3 \text{ kg} (10^3 \text{ J kg}^{-1} \text{ K}^{-1})}$$

$$\frac{dT}{dt} = 4 \times 10^{-3} \text{ K/s} = 14 \text{ K/hr}$$

(3)

$$3) \quad a) \quad B_\nu(\nu, T) = \frac{2h\nu^3}{c^2} \frac{1}{e^{h\nu/KT} - 1}$$

$$\text{If } h\nu/KT \ll 1 \quad \& \quad e^x = 1 + x + \frac{x^2}{2!} + \frac{x^3}{3!} + \dots$$

$$e^x \approx 1 + x \quad \text{for } x \ll 1$$

$$\Rightarrow B_\nu(\nu, T) = \frac{2h\nu^3}{c^2} \left(\frac{h\nu}{KT} \right)^{-1} = \frac{2KT\nu^2}{c^2}$$

$$\Rightarrow F = \int_{\nu_1}^{\nu_2} B_\nu d\nu = \frac{2KT\nu^2}{c^2} d\nu$$

$$\Rightarrow \boxed{F = \frac{2KT}{3c^2} (\nu_2^3 - \nu_1^3)} \quad \text{in } \text{W m}^2 \text{sr}^{-1}$$

b) Integrating over all direction and over the total surface, gives the power

$$P = \pi F A$$

$$\Rightarrow P = \frac{2\pi KT(4\pi a_\nu^2)}{3c^2} (\nu_2^3 - \nu_1^3)$$

$$= \frac{8\pi^2 K T a_\nu^2}{3c^2} \left(\frac{c^3}{\lambda_1^3} - \frac{c^3}{\lambda_2^3} \right)$$

$$= \frac{8\pi^2 k T a_v^2 c}{3} \left[\frac{1}{\lambda_1^3} - \frac{1}{\lambda_2^3} \right]$$

$$\approx \frac{8\pi^2 k T a_v^2 c}{3 \lambda_1^3} \quad \text{since } \lambda_2 = 100 \lambda_1$$

$$P \approx \frac{8\pi^2 (1.38 \times 10^{-23}) (737) (6.0 \times 10^6)^2 (3.0 \times 10^8)}{3 (1.0 \times 10^{-3})^3}$$

$$P = 2.9 \times 10^{12} \text{ W} = 2.9 \text{ TW}$$

c) at a distance $r = 41 \times 10^6 \text{ km}$

$$F = \frac{P}{4\pi r^2} = \frac{2.9 \times 10^{12} \text{ W}}{4\pi (41 \times 10^9)^2 \text{ m}^2}$$

$$F = 1.4 \times 10^{-10} \text{ W/m}^2 \quad \text{not too different from flux from cell phone}$$

with a 100 m^2 antenna $\Rightarrow$ $P = 14 \text{ nW}$