



ESS2222H

Tectonics and Planetary Dynamics
Lecture 10
Thermal Convection
Navier-Stokes Equations

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Thermal Convection

Conservation of Fluid in 2D

The flow entering a small volume $\delta x \delta y$ at x in $x - dir$: $\rho u \delta y$

The flow leaving a small volume $\delta x \delta y$ at $x + \delta x$ in $x - dir$: $\rho u(x + \delta x) \delta y$

The flow rate per unit area: $\rho u(x + \delta x)$

$$f(x) \approx f(a) + \frac{f'(a)}{1!} (x - a) + \frac{f''(a)}{2!} (x - a)^2 + \dots \text{ Taylor expansion}$$

$$f(x + \delta x) \approx f(x) + \frac{f'(x)}{1!} (x + \delta x - x) + \dots$$

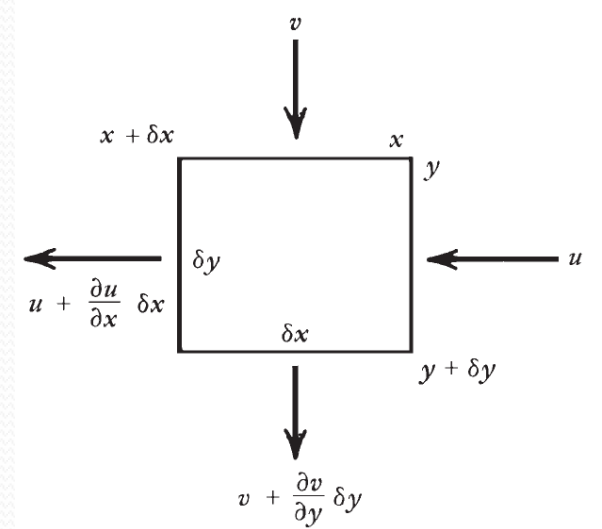
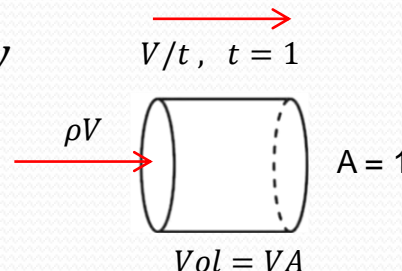
$$u(x + \delta x) = u(x) + \frac{\partial u}{\partial x} \delta x$$

The net flow passing through the volume (entering and leaving) in x-direction:

$$\rho \left[u(x) + \frac{\partial u}{\partial x} \delta x - u(x) \right] \delta y = \rho \frac{\partial u}{\partial x} \delta x \delta y$$

Similarly in $y - dir$:

$$\rho \left[v(x) + \frac{\partial v}{\partial y} \delta y - v(y) \right] = \rho \frac{\partial v}{\partial y} \delta y \delta x$$



Flow across the surfaces of an infinitesimal rectangular element.

Conservation of Fluid in 2D

The total net flow through the volume:

:

$$\rho \frac{\partial u}{\partial x} \delta x \delta y + \rho \frac{\partial v}{\partial y} \delta y \delta x = \rho \left[\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right] \delta y \delta x$$

If the flow is steady (time-independent), and there are no density variations, then there will be no net flow into or out of the volume.

The conservation of fluid or continuity equation is:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad \text{or} \quad \text{div } V = \nabla \cdot V = 0 \quad \text{Mass Conservation}$$

$$\text{In 3D:} \quad \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0, \quad V \equiv V(u, v, w)$$

In spherical coordinate :

$$\nabla \cdot V = \frac{1}{r^2} \frac{\partial}{\partial r} (r^2 V_r) + \frac{1}{r \sin(\theta)} \frac{\partial}{\partial \theta} (V_\theta \sin(\theta)) + \frac{1}{r \sin(\theta)} \frac{\partial V_\phi}{\partial \phi}$$

Momentum Equation

Elemental Force Balance

The forces acting on the small volume element are:

- 1) Pressure force
- 2) Viscous force
- 3) Gravity force (body force)
- 4) Inertial force

The Earth's mantle behave as a **highly viscous fluid** on geologic time scales.

The viscosity of mantle is $\sim \mu = 10^{21} \text{ Pa}\cdot\text{s}$

Its density is $\sim \rho = 4000 \text{ kg/m}^3$

And its thermal diffusivity is $\sim \kappa = 10^{-6} \text{ m}^2/\text{s}$

The Prandtl number: $Pr = \frac{\nu}{\kappa}$

$\kappa = \frac{K}{\rho C_P} \text{ m}^2/\text{s}$ *thermal diffusivity*

$\nu = \frac{\mu}{\rho} \text{ m}^2/\text{s}$ *kinematic viscosity*

μ *Pa}\cdot\text{s}* *dynamic viscosity*

K *W/(mK)* *thermal conductivity*

C_P *J/(kgK)* *specific heat*

ρ *kg / m}^3* *density*

$$Pr_{Earth} \sim 10^{23}$$

Momentum Equation

At high Prandtl numbers the inertial forces can be neglected:

$$\frac{\partial v}{\partial t} \approx 0$$

$$F_P + F_{visc} + F_g = 0$$

The force acting at x in $x - dir$ on $\delta y - element$: $p(x) \delta y$

The force acting at $x + \delta x$ in $x - dir$ on $\delta y - element$: $p(x + \delta x) \delta y$

The net pressure force on the element in the x -dir. per unit area of the fluid element:

$$\frac{p(x)\delta y - p(x+\delta x)\delta y}{\delta x \delta y} = - \frac{p(x+\delta x) - p(x)}{\delta x}$$

By virtue of a simple Taylor series expansion $\rightarrow - \frac{\partial p}{\partial x}$

Similarly for the pressure force on the element in the y -dir. Per unit area of the element: $-\frac{\partial p}{\partial y}$

Momentum Equation

The gravitational force acting on the volume element:

$$F_g = mg = \rho \delta x \delta y g$$

$$F_g = \rho g \quad \text{the net } g - \text{force per unit volume}$$

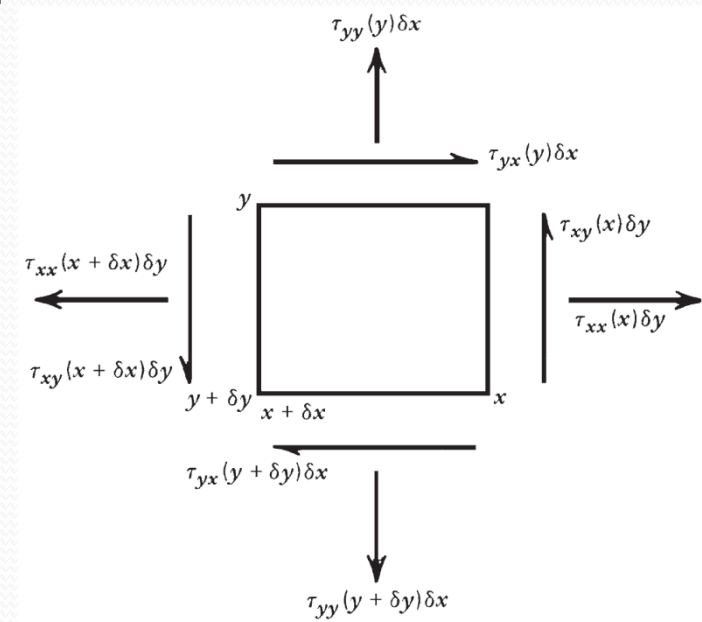
Viscus Foces

The net viscous force in the x-dir. Per unit volume:

$$\frac{\tau_{xx}(x+\delta x)\delta y - \tau_{xx}(x)\delta y}{\delta x \delta y} + \frac{\tau_{yx}(y+\delta y)\delta x - \tau_{yx}(y)\delta x}{\delta x \delta y}$$

Expanding $\tau_{xx}(x + \delta x)$ and $\tau_{yx}(y + \delta y)$ around x and y , respectively (using Taylor series expansion):

$$\frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yx}}{\partial y}$$



Momentum Equation

Similarly, the net viscous force in the y-dir. Per unit volume of the element:

$$\frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{xy}}{\partial x}$$

For an ideal Newtonian viscous fluid, the viscous stresses are linearly proportional to the velocity gradients:

$$\tau_{xx} \equiv \tau_{11} = 2\mu \frac{\partial u}{\partial x}, \quad \tau_{yy} \equiv \tau_{22} = 2\mu \frac{\partial v}{\partial y}$$

$$\tau_{xy} \equiv \tau_{12} = 2\mu \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right), \quad \tau_{xy} = \tau_{yx}$$

In general:

$$\tau_{ij} = 2\mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

where μ is the dynamic viscosity.

Momentum Equation

The application of

$$\left\{ \begin{array}{l} \tau_{xx} \equiv \tau_{11} = 2\mu \frac{\partial u}{\partial x} \\ \tau_{yy} \equiv \tau_{22} = 2\mu \frac{\partial v}{\partial y} \\ \tau_{xy} \equiv \tau_{12} = 2\mu \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \end{array} \right. \quad \text{in} \quad \left\{ \begin{array}{l} \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yx}}{\partial y} \\ \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{xy}}{\partial x} \end{array} \right. \begin{array}{l} \text{the net viscous force in the } x\text{-dir. Per unit volume} \\ \text{the net viscous force in the } y\text{-dir. Per unit volume} \end{array}$$

yields to;

$$2\mu \frac{\partial^2 u}{\partial x^2} + \mu \left(\frac{\partial^2 u}{\partial y^2} + \frac{\partial^2 v}{\partial x \partial y} \right) \text{ and } 2\mu \frac{\partial^2 v}{\partial y^2} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 u}{\partial x \partial y} \right), \text{ respectively.}$$

These expressions can be simplified using the continuity equation:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad \rightarrow \quad \frac{\partial^2 v}{\partial x \partial y} = -\frac{\partial^2 u}{\partial x^2}, \quad \frac{\partial^2 u}{\partial x \partial y} = -\frac{\partial^2 v}{\partial y^2}$$

$$2\mu \frac{\partial^2 u}{\partial x^2} + \mu \left(\frac{\partial^2 u}{\partial y^2} + \frac{\partial^2 v}{\partial x \partial y} \right) \rightarrow \mu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right)$$

$$2\mu \frac{\partial^2 v}{\partial y^2} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 u}{\partial x \partial y} \right) \rightarrow \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right)$$

Momentum Equation

The force balance equations for an incompressible fluid ($\nabla \cdot V = 0$):

$$-\frac{\partial p}{\partial x} + \mu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) = 0$$

$$-\frac{\partial p}{\partial y} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) + \rho g = 0$$

To **eliminate the hydrostatic** pressure variation in Equation:

$$P = p - \rho g y$$

The pressure P is the pressure generated by fluid flow. With this substitution:

$$-\frac{\partial P}{\partial x} + \mu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) = 0$$

$$-\frac{\partial P}{\partial y} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) = 0$$

In compact form and in 3D-geometry:

$$-\nabla P + \nabla^2 V = 0 \quad \text{Momentum Conservation}$$

Momentum Equation

The continuity equation for compressible fluid

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0 \quad \text{Mass Conservation}$$

U: Characteristic Vel. **C: Sound Vel.**

Gough (1969)

$$\text{If } M^2 = \frac{U^2}{C^2} \leq 1 \rightarrow \frac{\partial}{\partial t} \rightarrow 0 \rightarrow \nabla \cdot (\rho V) = 0 \quad M: \text{Mach number}$$

$$\nabla \cdot (\rho V) = \rho \nabla \cdot V + V \nabla \rho$$

Momentum Equation in More General Form

$$F_{inertial} = F_P + F_{visc} + F_g$$

$$F_{inertial} = \rho \left(\frac{\partial V}{\partial t} + V \nabla \cdot V \right)$$

$$\rho \left(\frac{\partial V}{\partial t} + V \nabla \cdot V \right) = -\nabla p + \nabla^2 V + \rho \bar{g}$$

The Stream Function

Incompressible 2D-Fluid

Solving the Momentum Equation Using Stream Function Method

Define $u = -\frac{\partial\psi}{\partial y}$, $v = +\frac{\partial\psi}{\partial x}$

Substituting in **continuity** equation:

$$-\frac{\partial^2\psi}{\partial x\partial y} + \frac{\partial^2\psi}{\partial y\partial x} = 0 \quad \text{which shows the stream function } \psi \text{ satisfies the continuity equation.}$$

Substituting in **momentum** equations:

$$\frac{\partial P}{\partial x} + \mu \left(\frac{\partial^3\psi}{\partial x^2\partial y} + \frac{\partial^3\psi}{\partial y^3} \right) = 0$$
$$-\frac{\partial P}{\partial y} + \mu \left(\frac{\partial^3\psi}{\partial x^3} + \frac{\partial^3\psi}{\partial y^2\partial x} \right) = 0$$

The Stream Function

For a single differential equation for ψ , the pressure can be eliminated from these equations by taking the partial derivative of these equations with respect to y and x , respectively:

Substituting in momentum equations:

$$\frac{\partial^4 \psi}{\partial x^4} + 2 \frac{\partial^4 \psi}{\partial x^2 \partial y^2} + \frac{\partial^4 \psi}{\partial y^4} = 0 \quad \text{biharmonic equation}$$

The Stream Function

Ex. – The flow accord AB?

The flow across AB can be calculated from the flows across AP and PB because conservation of mass.

The volumetric flow rate across AP into the triangle per unit distance normal to the figure is: $u\delta y$

similarly the flow rate across PB out of the triangle is: $v\delta x$

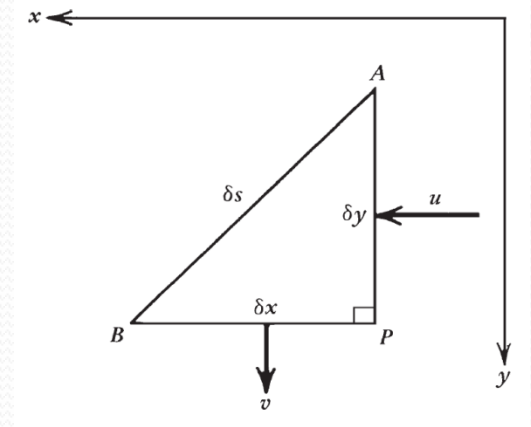
The net flow rate out of PAB is thus: $-u\delta y + v\delta x$

This must be equal to the volumetric flow rate into PAB across AB .

$$u = -\frac{\partial\psi}{\partial y}$$

$$v = +\frac{\partial\psi}{\partial x} \quad \rightarrow \quad -u\delta y + v\delta x = \frac{\partial\psi}{\partial y}\delta y + \frac{\partial\psi}{\partial x}\delta x \equiv d\psi(x, y) \text{ the volumetric flow rate between A and B}$$

$$\int_A^B d\psi = \psi_A - \psi_B \quad \text{for A and B at arbitrary distance}$$



Thermal Convection

Plate tectonics is the consequence of **thermal convection** in the mantle, driven largely by **radiogenic** heat sources and the **cooling** of the Earth.

Thermal convection is the consequence of a change in density by a change in temperature (thermal expansion). This situation is **gravitationally unstable** and the cool fluid tends to **sink** and the hot fluid **rises**.

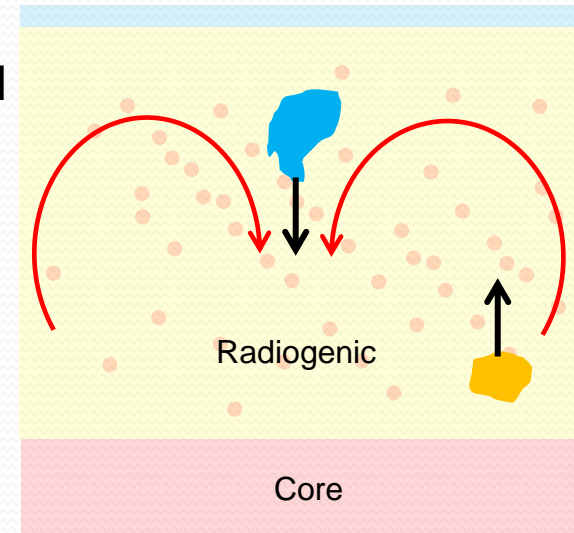
Density variations caused by thermal expansion lead to the **buoyancy forces** that drive thermal convection.

$$\rho = \rho_0 + \rho' \quad \rightarrow \quad \rho' = \rho - \rho_0 \quad F_B = \rho' g$$

$$F_g = \rho_0 g \quad \rightarrow \quad F_g + F_B$$

$$\rho_0 g \quad \rightarrow \quad \rho_0 g + \rho' g \quad \text{in momentum conservation equation}$$

$$\rho' \ll \rho_0 \quad \rho_0: \text{reference density}$$



Thermal Convection

In all other respects, however, the density variations are sufficiently small so that they can be neglected. This is known as the ***Boussinesq approximation***.

It allows us to use the **incompressible** conservation of fluid equation:

$$\nabla \cdot V = 0$$

$$-\frac{\partial p}{\partial x} + \mu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) = 0$$

$$-\frac{\partial p}{\partial y} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) + \rho g = 0 \quad \rightarrow \quad -\frac{\partial p}{\partial y} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) + \rho g + \rho' g = 0$$

Eliminating the hydrostatic pressure by introducing:

$$P = p - \rho_0 g y$$

Thermal Convection

$$-\frac{\partial P}{\partial x} + \mu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) = 0$$

$$-\frac{\partial P}{\partial y} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) + \rho' g = 0$$

$$\rho' = -\rho_0 \alpha (T - T_0)$$

α : volumetric coefficient of thermal expansion

T_0 : reference temperature

$$-\frac{\partial P}{\partial y} + \mu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) - \underbrace{g\rho_0\alpha(T - T_0)} = 0$$

Buoyancy force per unit volume

To find the velocity field from momentum conservation equations, we **need the temperature T**. Therefore we require the **heat equation (energy equation)** that governs the variation of temperature.

Heat transfer = Conduction + Convection

Thermal Convection

Thermal energy per unit volume: ρcT (ρcTu energy flux or energy flow per unit area)

Amount of heat transported across δy at x : $\rho cTu\delta y$ per unit time, crossing δy

Heat flux at $x + \delta x$: $\rho cTu + \frac{\partial(\rho cTu)}{\partial x} \delta x$

The **net energy advected** out of the elemental volume per unit time and per unit depth due to flow in the x direction is thus:

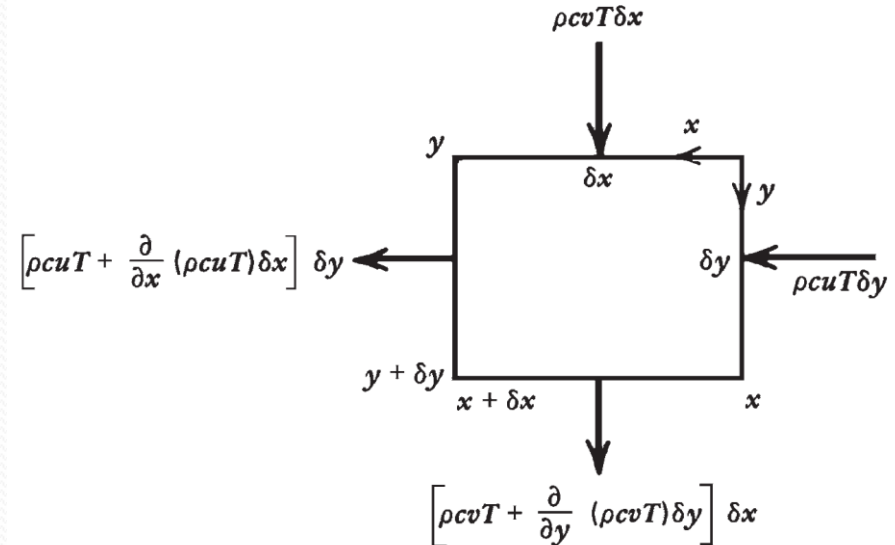
$$\left[\left[\rho cTu + \frac{\partial(\rho cTu)}{\partial x} \delta x \right] - \rho cTu \right] \delta y$$

$$= \frac{\partial(\rho cTu)}{\partial x} \delta x \delta y$$

Similarly in y direction:

$$\left[\left[\rho cTv + \frac{\partial(\rho cTv)}{\partial y} \delta y \right] - \rho cTv \right] \delta x$$

$$= \frac{\partial(\rho cTv)}{\partial y} \delta x \delta y$$



Thermal Convection

The **net rate of heat advection out of the element** by flow in both directions is:

$$\left[\frac{\partial(\rho c T u)}{\partial x} + \frac{\partial(\rho c T v)}{\partial y} \right] \delta x \delta y$$

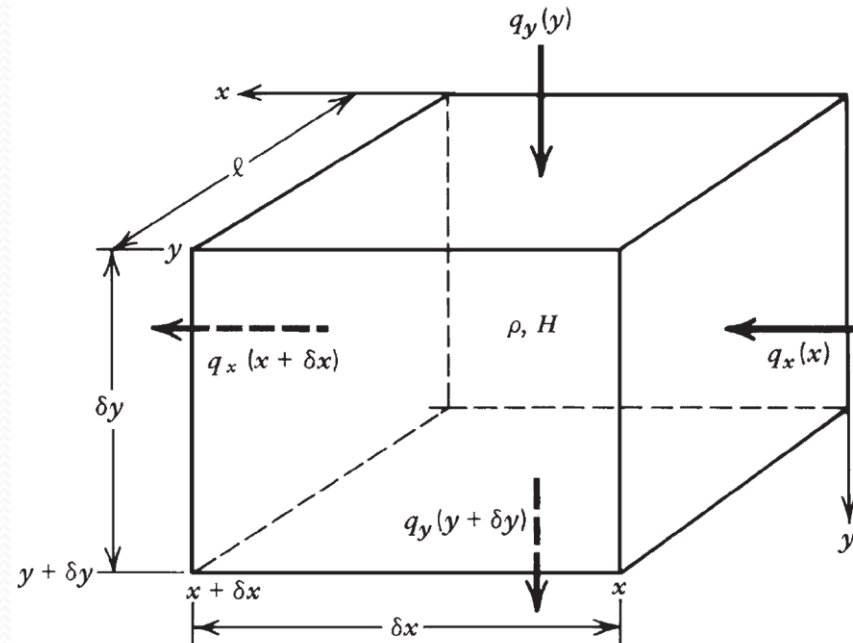
Heat Conduction

Heat flux in x direction at x : $q_x(x)$

Heat flux in x direction at $x + \delta x$: $q_x(x + \delta x)$

Heat flux in y direction at y : $q_y(y)$

Heat flux in y direction at $y + \delta y$: $q_y(y + \delta y)$



The net heat flow rate out of the element is:

$$\begin{aligned} & [q_x(x + \delta x) - q_x(x)] \delta y + [q_y(y + \delta y) - q_y(y)] \delta x \\ &= \frac{\partial q_x}{\partial x} \delta x \delta y + \frac{\partial q_y}{\partial y} \delta x \delta y = \left(\frac{\partial q_x}{\partial x} + \frac{\partial q_y}{\partial y} \right) \delta x \delta y \quad \text{using Taylor expansion} \end{aligned}$$

Thermal Convection

Steady State

In steady state

$$\frac{\partial q_x}{\partial x} + \frac{\partial q_y}{\partial y} = 0$$

In the presence of internal heating the rate of heat generation in the element is:
 $\rho H \delta x \delta y$

and

$$\frac{\partial q_x}{\partial x} + \frac{\partial q_y}{\partial y} = \rho H \quad \text{and in 3D:} \quad \frac{\partial q_x}{\partial x} + \frac{\partial q_y}{\partial y} + \frac{\partial q_z}{\partial z} = \rho H$$

Fourier's Law of Conduction

$$q_x = -k \frac{\partial T}{\partial x}, \quad q_y = -k \frac{\partial T}{\partial y}, \quad q_z = -k \frac{\partial T}{\partial z} \quad \text{for isotropic medium}$$

$$\rightarrow -k \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} \right) = \rho H \quad \text{or} \quad -k \nabla^2 T = \rho H$$

Thermal Convection

We had obtained the advection term:

$$\left[\frac{\partial(\rho c T u)}{\partial x} + \frac{\partial(\rho c T v)}{\partial y} \right] \delta x \delta y$$

And the conduction term:

$$-k \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right) \delta x \delta y$$

Energy Conservation

The **combined transport** of energy out of the elemental volume by conduction and convection must be **balanced** by the **change in the energy content** of the element. The thermal energy of the fluid is $\rho c T$ **per unit volume**. Thus, this quantity changes at the rate:

$$\frac{\partial(\rho c T)}{\partial t} \delta x \delta y$$

Thermal Convection

By combining the effects of conduction, convection, and *thermal inertia*, we obtain:

$$\underbrace{\frac{\partial(\rho c T)}{\partial t}}_{\text{Rate of change in heat}} - \underbrace{k \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right)}_{\text{Conduction}} + \underbrace{\frac{\partial(\rho c T u)}{\partial x} + \frac{\partial(\rho c T v)}{\partial y}}_{\text{Convection}} = 0 \quad \text{energy balance}$$

Rate of change
in heat

Conduction

Convection

For constant ρ and c and noting that

$$\frac{\partial(Tu)}{\partial x} + \frac{\partial(Tv)}{\partial y} = u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} + T \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) = u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y}$$

$$\nabla \cdot V = 0$$

The thermal diffusion is defined as:

$$\kappa = \frac{K}{\rho c}$$

$$\frac{\partial T}{\partial t} + u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \kappa \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right)$$

$$\frac{\partial T}{\partial t} + V \cdot \nabla T = \kappa \nabla^2 T$$

Thermal Convection

In this derivation we have neglected:

- a) frictional heating in the fluid associated with the resistance to flow
- b) compressional heating associated with the work done by pressure forces in moving the fluid



Navier-Stokes Equations

Navier-Stokes Equations

We saw that the **force balance** on an small volume element of fluid leads to the equation for **conservation of momentum**:

$$F_P + F_{visc} + F_g = 0$$

$$-\frac{\partial p}{\partial x_i} + -\frac{\partial \tau_{ij}}{\partial x_j} + \rho g_i = 0 \quad i^{th} \text{ component}$$

According to Newton's second law of motion, any **imbalance of forces** on the fluid parcel results in an **acceleration** of the elemental parcel:

$$\underbrace{\rho \frac{Du_i}{Dt}}_{\text{Inertial}} = -\underbrace{\frac{\partial p}{\partial x_i}}_{\text{Surface}} + \underbrace{\frac{\partial \tau_{ij}}{\partial x_j}}_{\text{Surface}} + \underbrace{\rho g_i}_{\text{Body forces}} = 0, \quad i = 1,2,3$$

Body forces:
 Gravity force
 Electromagnetic force
 Centrifugal force
 Coriolis force

where $\frac{Du}{Dt} = \frac{\partial u}{\partial t} + \frac{\partial u}{\partial x_i} \frac{dx_i}{dt} \equiv \frac{\partial u}{\partial t} + (u \cdot \nabla)u$ (total derivative)

$$F = ma \quad \rightarrow \quad \rho \frac{Du_i}{Dt} \quad \text{mass} \times \text{acceleration}$$

Navier-Stokes Equations

$$\nabla f = \frac{\partial f}{\partial x_1} \hat{i} + \frac{\partial f}{\partial x_2} \hat{j} + \frac{\partial f}{\partial x_3} \hat{k}$$

Note that:

$$\frac{Df}{Dt} = \frac{\partial f}{\partial t} + \frac{\partial f}{\partial x_i} \frac{dx_i}{dt} \equiv \frac{\partial f}{\partial t} + u \cdot \nabla f \quad \text{for scalar function } f$$

$$\frac{Du}{Dt} = \frac{\partial u}{\partial t} + \frac{\partial u}{\partial x_i} \frac{dx_i}{dt} \equiv \frac{\partial u}{\partial t} + (u \cdot \nabla)u \quad \text{for the velocity vector } u$$

$$\frac{DF}{Dt} = \frac{\partial F}{\partial t} + \frac{\partial F}{\partial x_i} \frac{dx_i}{dt} \equiv \frac{\partial F}{\partial t} + (u \cdot \nabla)F \quad \text{for vector function } F$$

Total derivative

Material derivative

Substantial derivative

Lagrangian derivative

Note also that:

$$\frac{\partial f}{\partial x_i} \frac{dx_i}{dt} = \frac{\partial f}{\partial x_i} u_i = \frac{\partial f}{\partial x_1} u_1 + \frac{\partial f}{\partial x_2} u_2 + \frac{\partial f}{\partial x_3} u_3 = \left(\frac{\partial f}{\partial x_1} \hat{i} + \frac{\partial f}{\partial x_2} \hat{j} + \frac{\partial f}{\partial x_3} \hat{k} \right) \cdot (u_1 \hat{i} + u_2 \hat{j} + u_3 \hat{k}) = \nabla f \cdot u$$

$$(u \cdot \nabla)F = \left(u_i \frac{\partial}{\partial x_i} \right) F \quad \rightarrow \quad (u \cdot \nabla) F_k = \left(u_i \frac{\partial}{\partial x_i} \right) F_k$$

Do not confuse total derivative with mass conservation

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0 \quad \text{Mass Conservation}$$

Navier-Stokes Equations

Also

$$\begin{aligned}(u \cdot \nabla)F &= \left[(u_1 \hat{i} + u_2 \hat{j} + u_3 \hat{k}) \cdot \left(\hat{i} \frac{\partial}{\partial x_1} + \hat{j} \frac{\partial}{\partial x_2} + \hat{k} \frac{\partial}{\partial x_3} \right) \right] (F_1 \hat{i} + F_2 \hat{j} + F_3 \hat{k}) = \\ & \left(u_1 \frac{\partial}{\partial x_1} + u_2 \frac{\partial}{\partial x_2} + u_3 \frac{\partial}{\partial x_3} \right) (F_1 \hat{i} + F_2 \hat{j} + F_3 \hat{k}) = \\ & \left(u_1 \frac{\partial F_1}{\partial x_1} + u_2 \frac{\partial F_1}{\partial x_2} + u_3 \frac{\partial F_1}{\partial x_3} \right) \hat{i} + \left(u_1 \frac{\partial F_2}{\partial x_1} + u_2 \frac{\partial F_2}{\partial x_2} + u_3 \frac{\partial F_2}{\partial x_3} \right) \hat{j} + \left(u_1 \frac{\partial F_3}{\partial x_1} + u_2 \frac{\partial F_3}{\partial x_2} + u_3 \frac{\partial F_3}{\partial x_3} \right) \hat{k}\end{aligned}$$

Navier-Stokes Equations

In the **absence of flow**, the only surface force is the **pressure force**:

$$-\frac{\partial p}{\partial x_i}$$

With flow, additional **deviatoric** forces act on the surface of an elemental parcel:

$$\frac{\partial \tau_{ij}}{\partial x_j} \sim \text{gradient of the velocities}$$

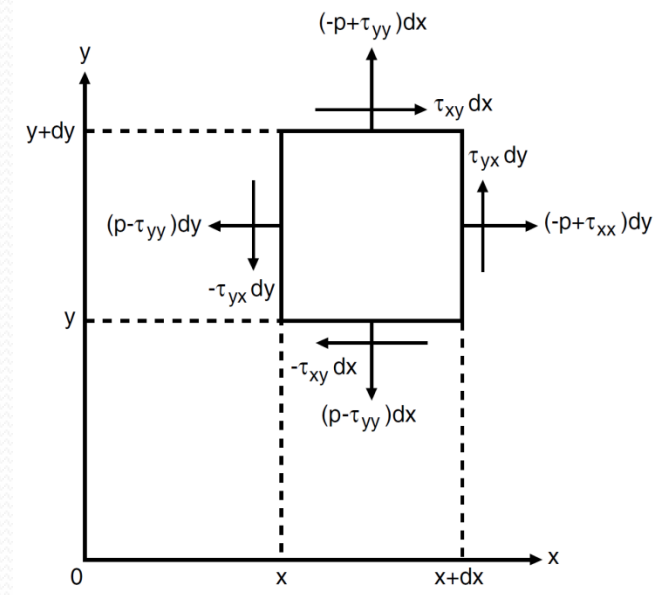
In 2D

$$\rho \left(\frac{\partial u_x}{\partial t} + u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} \right) = -\frac{\partial p}{\partial x} + \frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{xy}}{\partial y}$$

$$\rho \left(\frac{\partial u_y}{\partial t} + u_x \frac{\partial u_y}{\partial x} + u_y \frac{\partial u_y}{\partial y} \right) = -\frac{\partial p}{\partial y} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{xy}}{\partial x} + \rho g$$

inertial term

$$\varepsilon_{ij} = \frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \quad \text{strain tensor}$$



Navier-Stokes Equations

Constitutive Law - Newtonian Fluid

Newtonian fluid is a fluid for which the **dependence** of τ_{ij} on ε_{ij} is **linear**.

In addition if the medium is also **isotropic** (the constants of proportionality in the deviatoric stress–strain rate relation are **independent of the orientation** of coordinate system axes), then

$$\tau_{ij} = 2\mu\varepsilon_{ij} + \lambda\varepsilon_{kk}\delta_{ij}$$

where μ is the **dynamic viscosity**, and λ is the **second viscosity**.

$$\rightarrow \frac{\tau_{ii}}{3} = \left(\lambda + \frac{2}{3}\mu\right)\varepsilon_{ii} \equiv k_B\varepsilon_{ii} \quad \rightarrow \quad \lambda = \left(k_B - \frac{2}{3}\mu\right) \quad (\delta_{ii} = \delta_{11} + \delta_{11} + \delta_{11} = 3)$$

where k_B is called **bulk viscosity**, a **measure of dissipation** under compression or expansion.

Navier-Stokes Equations

Combining these two equations:

$$\tau_{ij} = 2\mu\varepsilon_{ij} + \left(k_B - \frac{2}{3}\mu\right)\varepsilon_{kk}\delta_{ij} \quad \rightarrow \quad \tau_{ij} = \mu\left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i}\right) + \left(k_B - \frac{2}{3}\mu\right)\frac{\partial u_k}{\partial x_k}\delta_{ij}$$

And the momentum equation:

$$\rho \frac{Du_i}{Dt} = -\frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) + \left(k_B - \frac{2}{3}\mu \right) \frac{\partial u_k}{\partial x_k} \delta_{ij} \right] + \rho g_i = 0$$

For many fluids k_B is very small. **$k_B \approx 0$ Stokes assumption**

Constitutive Law with $k_B \approx 0$

With $k_B \approx 0$, the constitutive or rheological law connecting deviatoric stress and strain rate becomes:

$$\tau_{ij} = 2\mu\varepsilon_{ij} - \frac{2}{3}\mu\varepsilon_{kk}\delta_{ij} \quad \rightarrow \quad \tau_{ij} = \mu\left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} - \frac{2}{3}\frac{\partial u_k}{\partial x_k}\delta_{ij}\right)$$

Navier-Stokes Equations

The Navier–Stokes equation

$$\rho \frac{Du_i}{Dt} = -\frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} - \frac{2}{3} \frac{\partial u_k}{\partial x_k} \delta_{ij} \right) \right] + \rho g_i = 0$$

Incompressible flow

For incompressible flow $\nabla \cdot u = \frac{\partial u_k}{\partial x_k} = 0$

$$\rho \frac{Du_i}{Dt} = -\frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \left[\mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \right] + \rho g_i = 0$$

If the dynamic viscosity (μ) is constant:

$$\rho \frac{Du_i}{Dt} = -\frac{\partial p}{\partial x_i} + \mu \left[\left(\frac{\partial^2 u_i}{\partial x_j^2} + \frac{\partial}{\partial x_i} \frac{\partial u_j}{\partial x_j} \right) \right] + \rho g_i = 0$$

$$\rho \frac{Du_i}{Dt} = -\frac{\partial p}{\partial x_i} + \mu \frac{\partial^2 u_i}{\partial x_j^2} + \rho g_i = 0$$

Energy Equation

For a simple case, in the **absence** of **internal heating** and **viscous dissipation**, and where the **density** and heat **capacity** were **constants**, we had obtained the energy balance equation as:

$$\frac{DT}{Dt} = \kappa \nabla^2 T \quad \text{or} \quad \frac{\partial T}{\partial t} + \mathbf{V} \cdot \nabla T = \kappa \nabla^2 T$$

In the presence of **viscous dissipation** and **internal heat sources**, and variable thermal conductivity, the energy conservation relation can be written as:

$$\underbrace{\rho c_p \frac{DT}{Dt}}_{\text{Rate of Change In energy}} = \underbrace{\nabla \cdot (K \nabla T)}_{\text{By conduction}} + \underbrace{\Phi}_{\text{By dissipation}} + \underbrace{\rho H}_{\text{By internal heating}} \quad \text{Energy balance}$$

Where H is the rate of internal heat production per unit mass and $\Phi \equiv \tau_{ij} \frac{\partial u_i}{\partial x_j}$ is viscous dissipative heat.

$$\frac{DT}{Dt} = \underbrace{\frac{\partial T}{\partial t}}_{\text{By time}} + \underbrace{\mathbf{u} \cdot \nabla T}_{\text{By advection}}$$

↑
Change in velocity

Navier-Stokes Equations

For a **Newtonian** fluid:

Since $\tau_{ij} = \mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) + \left(k_B - \frac{2}{3} \mu \right) \frac{\partial u_k}{\partial x_k} \delta_{ij}$, then

$$\Phi = \left[\mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) + \left(k_B - \frac{2}{3} \mu \right) \frac{\partial u_k}{\partial x_k} \delta_{ij} \right] \frac{\partial u_i}{\partial x_j}$$
$$\rightarrow \Phi = k_B \left(\frac{\partial u_k}{\partial x_k} \right)^2 + 2\mu \left[\frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) - \frac{1}{3} \left(\frac{\partial u_k}{\partial x_k} \right) \delta_{ij} \right]^2$$

If the fluid is incompressible:

$$\Phi = \frac{\mu}{2} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)^2$$

- The **bulk viscosity** $k_B = \left(\lambda + \frac{2}{3} \mu \right)$ leads to dissipation due to **volume changes** in a deforming fluid.
- The dynamic viscosity μ leads to dissipation **through shear**. Note that there are **no volume changes** associated with the bracketed tensor in the second term.

$$\text{i.e., } \frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) - \frac{1}{3} \left(\frac{\partial u_k}{\partial x_k} \right) \delta_{ij} = 0 \text{ for } i=j$$

Navier-Stokes Equations

It can be shown that for **compressible flow**:

$$\rho c_P \frac{DT}{Dt} - \alpha T \frac{DP}{Dt} = \nabla \cdot (K \nabla T) + \Phi + \rho H$$

$$\rho c_P \left(\frac{\partial T}{\partial t} + u \cdot \nabla T \right) - \alpha T \left(\frac{\partial P}{\partial t} + u \cdot \nabla P \right) = \nabla \cdot (K \nabla T) + \Phi + \rho H$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot \left(\nabla T - \frac{\alpha T}{\rho c_P} \nabla P \right) \right] = \nabla \cdot (K \nabla T) + \Phi + \rho H$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \Phi + \rho H$$

where $\nabla T_S \equiv \frac{\alpha T}{\rho c_P} \nabla P$ *Adiabatic temperature gradient*

$$\nabla P = 0\hat{i} + 0\hat{j} + \frac{\partial P}{\partial z} \hat{k} \approx \frac{\partial(-\rho g z)}{\partial z} \hat{k} \rightarrow \nabla T_S \approx -\frac{g\alpha}{c_P} T$$

and $\Phi \equiv \tau_{ij} \frac{\partial u_i}{\partial x_j}$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \tau_{ij} \frac{\partial u_i}{\partial x_j} + \rho H$$

Basic Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0$$

$$\rho \left(\frac{\partial u_i}{\partial t} + (u \cdot \nabla) u_i \right) = - \frac{\partial p}{\partial x_i} + \frac{\partial \tau_{ij}}{\partial x_j} + \rho g_i = 0$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \tau_{ij} \frac{\partial u_i}{\partial x_j} + \rho H$$

$$\rho = \rho_r \left[1 - \alpha(T - T_r) + \frac{1}{K_T} (P - P_r) \right] + \Delta \rho_i (\Gamma_i - \Gamma_{ri}) \quad i=1,2,3$$

$$\Gamma_i = \frac{1}{2} [1 + \tanh(\pi_i)]$$

$$\pi_i = \frac{d_i - d - \gamma_i (T - T_i)}{h_i}$$

State Equation

$$\rho(P, T) = ?$$

Taylor expansion

$$\rho(P, T) = \rho_r + \left(\frac{\partial \rho_r}{\partial T}\right)_P (T - T_r) + \left(\frac{\partial \rho_r}{\partial P}\right)_T (P - P_r) \quad \text{where} \quad \rho_r \equiv \rho(P_r, T_r)$$

P_r : hydrostatic pressure

T_r : adiabatic temperature

But

$$\alpha = \frac{1}{\rho_r} \left(\frac{\partial \rho_r}{\partial T}\right)_P, \quad \text{and} \quad K_T = \rho_r \left(\frac{\partial P}{\partial \rho_r}\right)_T$$

$$\rho(P, T) = \rho_r \left[1 - \alpha(T - T_r) + \frac{(P - P_r)}{K_T} \right]$$

State Equation

$$\nabla \rho_r(S, P_H) = \left(\frac{\partial \rho_r}{\partial S} \right)_{P_H} \nabla S + \left(\frac{\partial \rho_r}{\partial P_H} \right)_S \nabla P_H$$

Since reference state is adiabatic: $\nabla S = 0$

and since $\nabla P_H = 0 + 0 + \frac{dP_H}{dz} \hat{k} = -\rho_r g \hat{k}$

and $\nabla \rho_r(S, P_H) = 0 + 0 + \frac{d\rho_r}{dz} \hat{k}$

$$\frac{1}{\rho_r} \frac{d\rho_r}{dz} = -g \left(\frac{\partial \rho_r}{\partial P_H} \right)_S = -\frac{g\rho_r}{K_S} \quad \text{where} \quad K_S = \rho_r \left(\frac{\partial P_H}{\partial \rho_r} \right)_S$$

Gruneisen's parameter is defined as: $\Gamma = \frac{\alpha K_S}{\rho C_P} = \frac{\alpha K_T}{\rho C_V}$

then:

$$\frac{1}{\rho_r} \frac{d\rho_r}{dz} = -\frac{g\alpha}{\Gamma C_P} = -\frac{1}{\Gamma H_T} \quad \text{where} \quad H_T = \frac{C_P}{g\alpha} \quad \text{scale height}$$

This is the well-known **Adams-Williamson** relation for a chemically homogeneous adiabatic density distribution under hydrostatic pressure (Birch 1952).

State Equation

Spiegel & Veronis (1960) gave criteria for the applicability of the **Boussinesq** approximation to compressible fluids:

$$\frac{d}{H_T} \ll 1 \quad \text{for shallow layers, } d: \text{characteristic length}$$

$$\text{For constant } \Gamma: \quad \frac{1}{\rho_r} \frac{d\rho_r}{dz} = -\frac{1}{\Gamma H_T} \quad \rightarrow \quad \int \frac{d\rho_r}{\rho_r} = -\int_d^z \frac{dz}{\Gamma H_T}$$

$$\ln \left[\frac{\rho_r(z)}{\rho_0(z)} \right] = \frac{d-z}{\Gamma H_T} \quad \rightarrow \quad \rho_r(z) = \rho_0 \exp(d-z)/\Gamma H_T$$

where $\rho_0 = \rho_r(z = d)$ is the density at the upper surface (*bottom: z=0*).

$$\Gamma \approx 1.1$$

$$\rho_r(z = 0) = \rho_0 \exp(d/\Gamma H_T)$$

$$\rho_r(z = d) = \rho_0$$

$$\Delta\rho \approx 0 \text{ if } \frac{d}{H_T} \ll 1$$

State Equation

We also have:

$$\frac{1}{K_T} = \frac{\alpha}{\rho_r C_V} \quad \text{and} \quad C_V = \frac{C_P}{1 + \alpha \Gamma T_r}$$

$$\left\{ \begin{array}{l} \rho(P, T) = \rho_r \left[1 - \alpha(T - T_r) + \frac{(P - P_r)}{K_T} \right] \\ \rho_r(z) = \rho_0 \exp(d - z) / \Gamma H_T \end{array} \right.$$

$$\rightarrow \rho(P, T) = \rho_r \left[1 - \alpha(T - T_r) + \alpha \left[(1 + \alpha \Gamma T_r) / (\Gamma \rho_r C_P) \right] (P - P_r) \right]$$

Note that $\rho_r = \rho_0 \exp(d - z) / \Gamma H_T$

In non-dimensional form $\rho_r' = \exp[(1 - z')D / \Gamma]$ where $D = d / H_T$

$$z' = z / d, \quad \rho' = \rho / \rho_0,$$

State Equation

The non-dimensional form can be written as:

$$\rho(P, T) = \rho_r [1 - \mu(T - T_r) + \mu D (1/(\Gamma \rho_r))(P - P_r) + \mu^2 D [\Gamma (T_r + T_0)/(\Gamma \rho_r)](P - P_r)]$$

$$\mu = \alpha \Delta T, D = d/H_T$$

For liquids: $\mu \ll 1$

For shallow depths: $D \ll 1$

For dilute gases: $\mu \approx 1$

For Boussinesq approximation: $\mu \ll 1, D \ll 1$

$$\rho(P, T) = \rho_r [1 - \mu(T - T_r)]$$

For anelastic liquid approximation: $\mu \ll 1, D \sim 1$

$$\rho(P, T) = \rho_r [1 - \mu(T - T_r) + \mu D (1/(\Gamma \rho_r))(P - P_r)]$$

Basic Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0$$

$$\rho \left(\frac{\partial u_i}{\partial t} + (u \cdot \nabla) u_i \right) = - \frac{\partial p}{\partial x_i} + \frac{\partial \tau_{ij}}{\partial x_j} + \rho g_i = 0$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \tau_{ij} \frac{\partial u_i}{\partial x_j} + \rho H$$

$$\rho(P, T) =$$

$$\rho_r \left[1 - \mu(T - T_r) + \mu D (1 / (\Gamma \rho_r)) (P - P_r) + \mu^2 D [\Gamma (T_r + T_0) / (\Gamma \rho_r)] (P - P_r) \right]$$

Anelastic Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0$$

$$\rho \left(\frac{\partial u_i}{\partial t} + (u \cdot \nabla) u_i \right) = - \frac{\partial p}{\partial x_i} + \frac{\partial \tau_{ij}}{\partial x_j} + \rho g_i = 0$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \tau_{ij} \frac{\partial u_i}{\partial x_j} + \rho H$$

$$\rho(P, T) =$$

$$\rho_r \left[1 - \mu(T - T_r) + \mu D (1/(\Gamma \rho_r)) (P - P_r) + \mu^2 D [\Gamma (T_r + T_0) / (\Gamma \rho_r)] (P - P_r) \right]$$

Anelastic Liquid Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0$$

$$\rho \left(\frac{\partial u_i}{\partial t} + (u \cdot \nabla) u_i \right) = - \frac{\partial p}{\partial x_i} + \frac{\partial \tau_{ij}}{\partial x_j} + \rho g_i = 0$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \tau_{ij} \frac{\partial u_i}{\partial x_j} + \rho H$$

$$\rho(P, T) =$$

$$\rho_r \left[1 - \mu(T - T_r) + \mu D (1/(\Gamma \rho_r)) (P - P_r) + \mu^2 D [\Gamma (T_r + T_0) / (\Gamma \rho_r)] (P - P_r) \right]$$

Truncated Anelastic Liquid Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0$$

$$\rho \left(\frac{\partial u_i}{\partial t} + (u \cdot \nabla) u_i \right) = - \frac{\partial p}{\partial x_i} + \frac{\partial \tau_{ij}}{\partial x_j} + \rho g_i = 0$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \tau_{ij} \frac{\partial u_i}{\partial x_j} + \rho H$$

$$\rho(P, T) =$$

$$\rho_r \left[1 - \mu(T - T_r) + \mu D (1/(\Gamma \rho_r)) (P - P_r) + \mu^2 D [\Gamma (T_r + T_0)/(\Gamma \rho_r)] (P - P_r) \right]$$

Extended Boussinesq Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0$$

$$\rho \left(\frac{\partial u_i}{\partial t} + (u \cdot \nabla) u_i \right) = - \frac{\partial p}{\partial x_i} + \frac{\partial \tau_{ij}}{\partial x_j} + \rho g_i = 0$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \tau_{ij} \frac{\partial u_i}{\partial x_j} + \rho H$$

$$\rho(P, T) =$$

$$\rho_r \left[1 - \mu(T - T_r) + \mu D (1/(\Gamma \rho_r)) (P - P_r) + \mu^2 D [\Gamma (T_r + T_0)/(\Gamma \rho_r)] (P - P_r) \right]$$

$$\rho_r = \rho_{surf} = \text{Const.}$$

Boussinesq Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho V) = 0$$

$$\rho \left(\frac{\partial u_i}{\partial t} + (u \cdot \nabla) u_i \right) = - \frac{\partial p}{\partial x_i} + \frac{\partial \tau_{ij}}{\partial x_j} + \rho g_i = 0$$

$$\rho c_P \left[\frac{\partial T}{\partial t} - \frac{\alpha T}{\rho c_P} \frac{\partial P}{\partial t} + u \cdot (\nabla T - \nabla T_S) \right] = \nabla \cdot (K \nabla T) + \tau_{ij} \frac{\partial u_i}{\partial x_j} + \rho H$$

$$\rho(P, T) =$$

$$\rho_r \left[1 - \mu(T - T_r) + \mu D (1/(\Gamma \rho_r)) (P - P_r) + \mu^2 D [\Gamma (T_r + T_0)/(\Gamma \rho_r)] (P - P_r) \right]$$

$$\rho_r = \rho_{surf} = \text{Const.}$$

Energy Regime

$$Q_{Surf} = Q_{Sec,Mant} + Q_{Rad} + Q_{CMB},$$

$$Q_{Sec,Mant} = \left(MC \frac{dT}{dt} \right)_{Mantle}$$

$$Q_{CMB} = Q_{Sec,Core} + Q_L + Q_G$$

$$Q_{Sec,Core} = \left(MC \frac{dT}{dt} \right)_{Core}$$

Q_{Surf} Surface heat flow (W)

$Q_{Sec,Mant}$ Secular cooling of mantle

Q_{Rad} Radiogenic heat

Q_{CMB} CMB heat flow

$Q_{Sec,Core}$ Secular cooling of core

Q_L Latent heat flow from the inner core boundary due to solidification

Q_G Gravitational heat flow from the inner core boundary due to solidification

Dimensionless Numbers

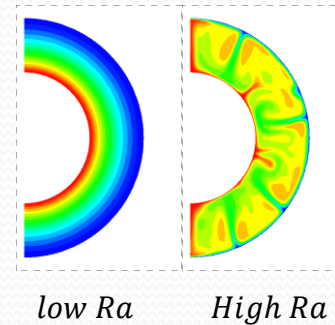
Rayleigh number

The Rayleigh number (Ra) for a fluid is a **dimensionless** number associated with buoyancy-driven flow that characterises the fluid's flow regime. **Lower** values denote **laminar** flow; and **higher** values denote **turbulent** flow. Below a **threshold** value (**critical Rayleigh number**), there is no fluid motion and heat transfer is by conduction rather than convection.

$$Ra_T = \frac{\rho g \alpha T d^3}{\nu K}$$

$$Ra_H = \frac{g \alpha H d^5}{\nu \kappa K}$$

$$Ra_T = 10^7 \text{ for mantle}$$



ρ : density (kg/m^3)

ν : Kinematic viscosity (m^2/s)

d : characteristic length (m)

g : gravity acc. (m/s^2)

T : temperature (Kelvin)

α : thermal expansivity ($1/\text{K}$)

K : thermal cond. ($\text{W}/\text{m}/\text{K}$)

κ : thermal diffusivity (m^2/s)

$\mu = \nu\rho$: dynamic viscosity ($\text{Pa} \cdot \text{s}$; $\text{N} \cdot \text{s}/\text{m}^2$; $\text{kg}/(\text{m} \cdot \text{s})$)

$$\kappa = \frac{K}{\rho C_P} \left(\frac{\text{W}/\text{m}/\text{K}}{(\text{kg}/\text{m}^3)(\text{J}/\text{kg}/\text{K})} = \text{m}^2/\text{s} \right)$$

Dimensionless Numbers

Prandtl number

The Prandtl number (Pr) is a dimensionless number, defined as the ratio of **momentum diffusivity** to **thermal diffusivity** (after the German physicist Ludwig Prandtl)

$$Pr = \frac{\nu}{\kappa} = \frac{\text{viscous diff. rate}}{\text{thermal diff. rate}} \quad Pr \sim \frac{10^{20}}{10^{-6}} = 10^{26} \text{ for mantle}$$

ν : Kinematic viscosity (m^2/s)

κ : thermal diffusivity (m^2/s) $\kappa = \frac{K}{\rho C_p} \left(\frac{W/m/K}{(kg/m^3)(J/kg/K)} \right) = m^2/s$

For mantle $\nu \gg \kappa$ $Pr \rightarrow \infty$

$Pr \ll 1$: the **thermal** diffusivity dominates

$Pr \gg 1$: the **momentum** diffusivity dominates

Ex. – In liquid mercury the **heat conduction** is more **significant** compared to **convection**.

For engine oil, **convection** is very **effective** in transferring **energy** (compared to pure conduction), so **momentum** diffusivity is **dominant**.

Dimensionless Numbers

Nusselt Number

The Nusselt number (Nu) is the ratio of convective to conductive heat transfer at a boundary in a fluid. **Convection** includes both **advection** (fluid motion) and **diffusion** (conduction).

$$Nu = \frac{h}{K/d} = \frac{\text{convective heat transfer coef.}}{\text{conductive heat transfer coef.}}$$

d: characteristic length (m)

h: convective heat transfer coef. ($h = \frac{q}{\Delta T}$)

q: heat flux (W/m^2)

ΔT : temperature difference (Kelvin)

Nu = 1, pure conduction

Nu = 1 – 10, slug flow

Nu = 100 – 1000, turbulent flow

Dimensionless Numbers

Reynolds Number

The Reynolds number (Re) is a **measure** of the **flow patterns** in a fluid. **Laminar** flow (sheet-like) has **low Reynolds** number, while **turbulent** flow has **higher** values of Reynolds number.

$$\begin{aligned} Re &= \frac{\text{inertial forces}}{\text{viscous forces}} = \frac{ma}{\tau A} = \frac{(\rho V) \cdot du/dt}{\mu du/dy \cdot A} = \frac{\rho d^3 \cdot du/dt}{\mu du/dy d^2} \\ &= \frac{\rho d \cdot dy/dt}{\mu} = \frac{\rho du}{\mu} = \frac{ud}{\nu} \end{aligned}$$

d: characteristic length (m)

u: velocity (m/s)

μ: dynamic viscosity (Pa · s)

Dimensionless Numbers

Peclet Number

The Peclet number is defined to be the ratio of the rate of advection to the rate of diffusion.

$$Pe = \frac{\text{rate of advection}}{\text{rate of diffusion}} = \frac{ud}{\kappa} = Re \times Pr$$

d: characteristic length (m)

u: velocity (m/s)

κ: thermal diffusivity (m²/s)

Dimensionless Numbers

Mach Number

The Mach number is ratio of convective velocity to sound velocity.

$$M = \frac{\text{convective velocity}}{\text{sound velocity}} = \frac{u}{c}$$

u : velocity (m/s)

c : velocity (m/s)

$M^2 \ll 1 \rightarrow$ a separation of time scales \rightarrow elastic vibrations irrelevant on convective time scales.